



University of  
**BRISTOL**



**AIRBUS**  
DEFENCE & SPACE

*An Introduction to the*  
**Template for Early Functional Definition and Analysis**

INCOSE UK MBSE Interest Group  
17<sup>th</sup> September 2019

Joe Gregory, Lucy Berthoud, Theo Tryfonas  
Ludovic Faure

University of Bristol, UK  
Airbus, UK

## But first...

- Last time I was here I asked some of you for help with an MBSE Questionnaire
- Work is ongoing
- Finishing off a paper on a previous set of interviews
- I will explain where this fits in



## MBSE Questionnaire

INCOSE UK MBSE Interest Group  
26<sup>th</sup> March 2019

Joe Gregory, Lucy Berthoud, Theo Tryfonas  
Ludovic Faure

University of Bristol, UK  
Airbus, UK

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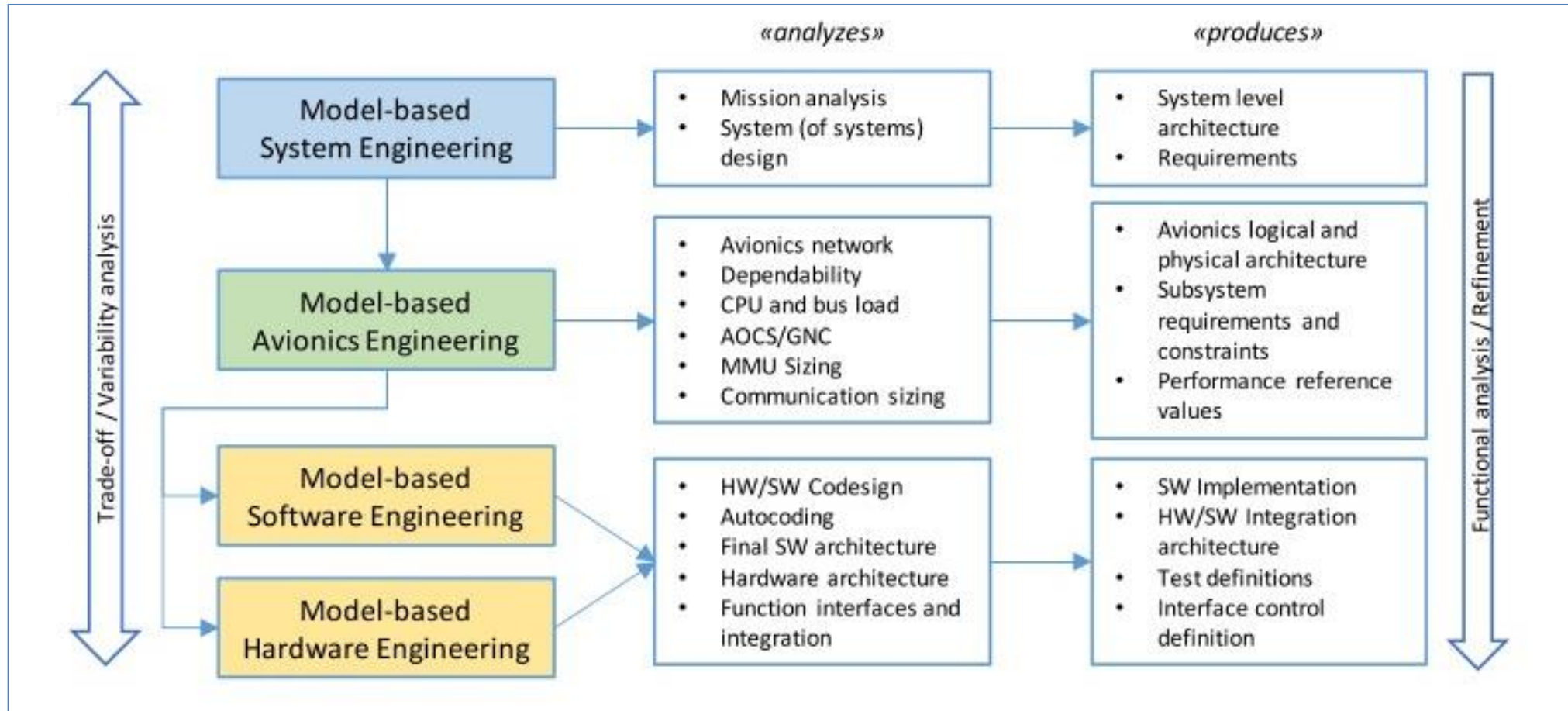
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3. Mass Memory Sizing
4. Critical Sequences
5. Template for Early Functional Definition and Analysis

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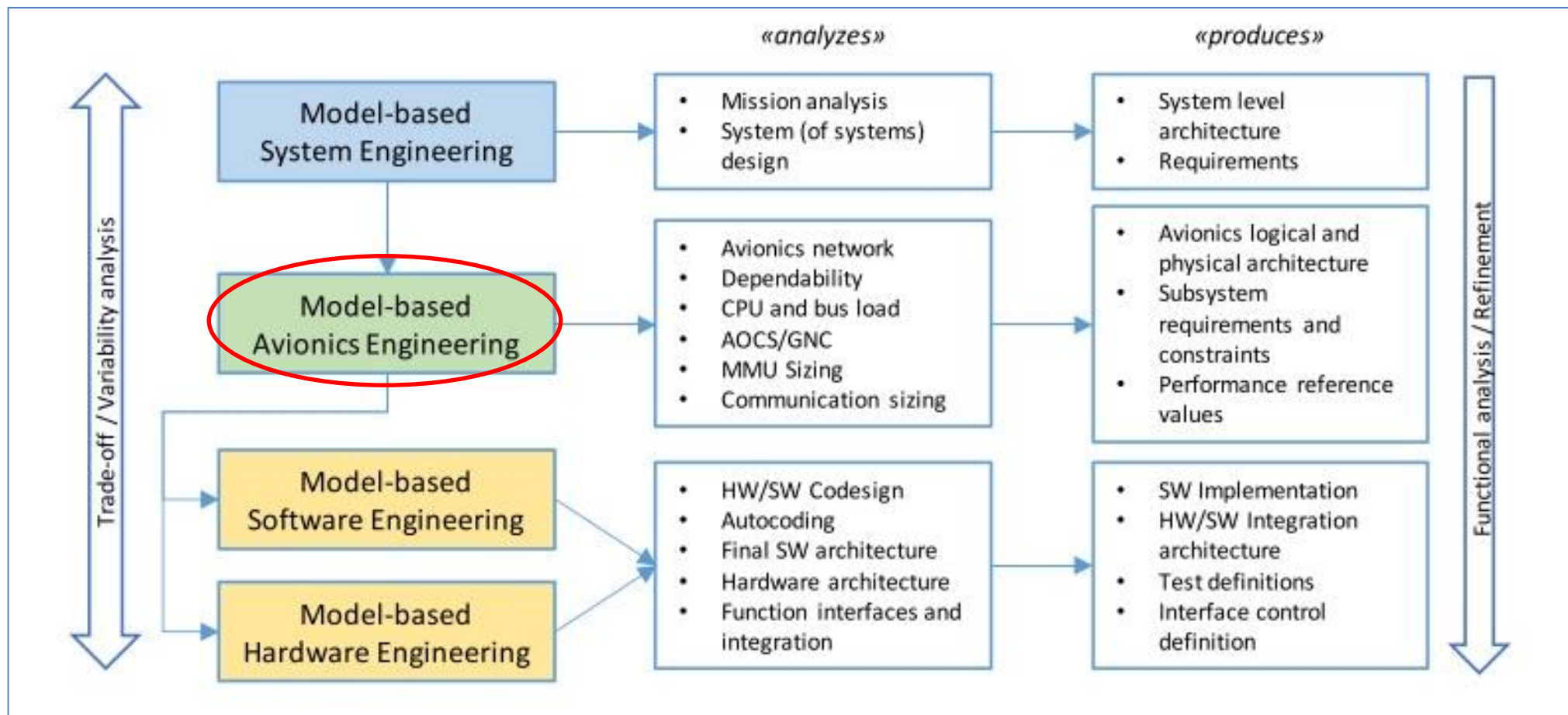
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## Functional Avionics – A quick look at MBAE

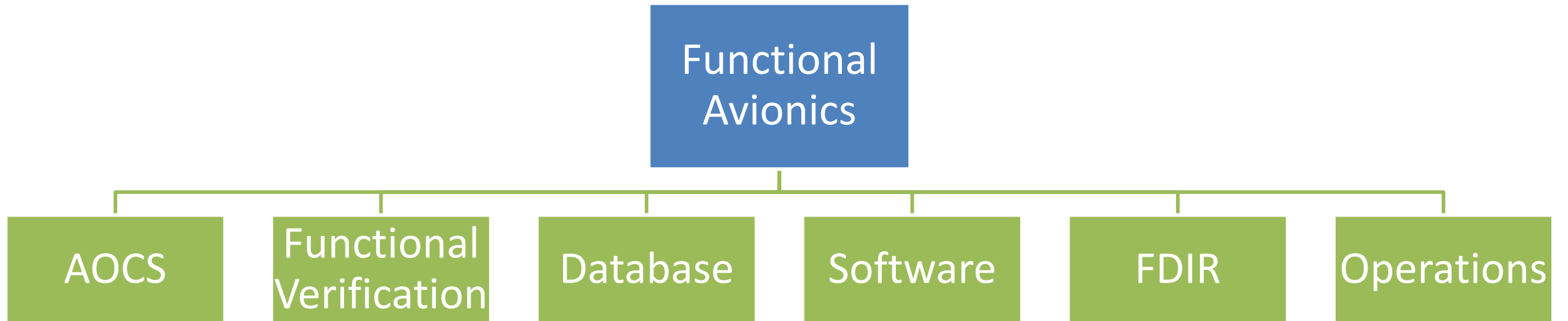


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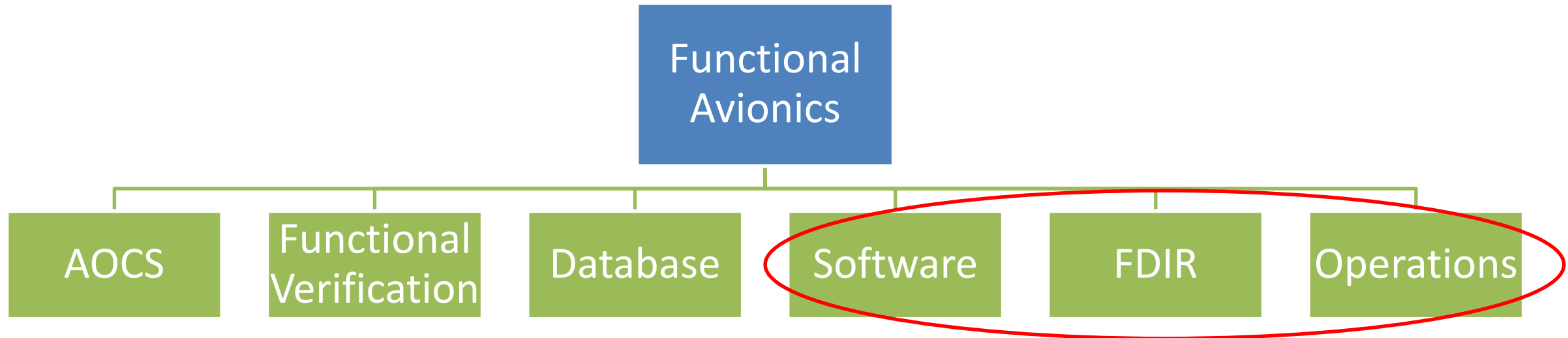


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## Airbus Space Functional Avionics: UK



## Airbus Space Functional Avionics: UK

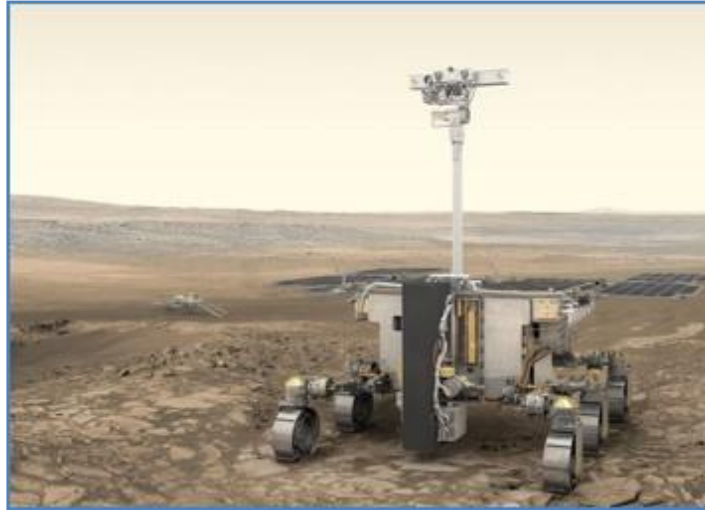




## Airbus Space MBSE Projects



- **eDeorbit**
- Bremen, Germany
- Autonomous fail-safe reaction



- **ExoMars Rover**
- Stevenage, UK
- Communication of requirements



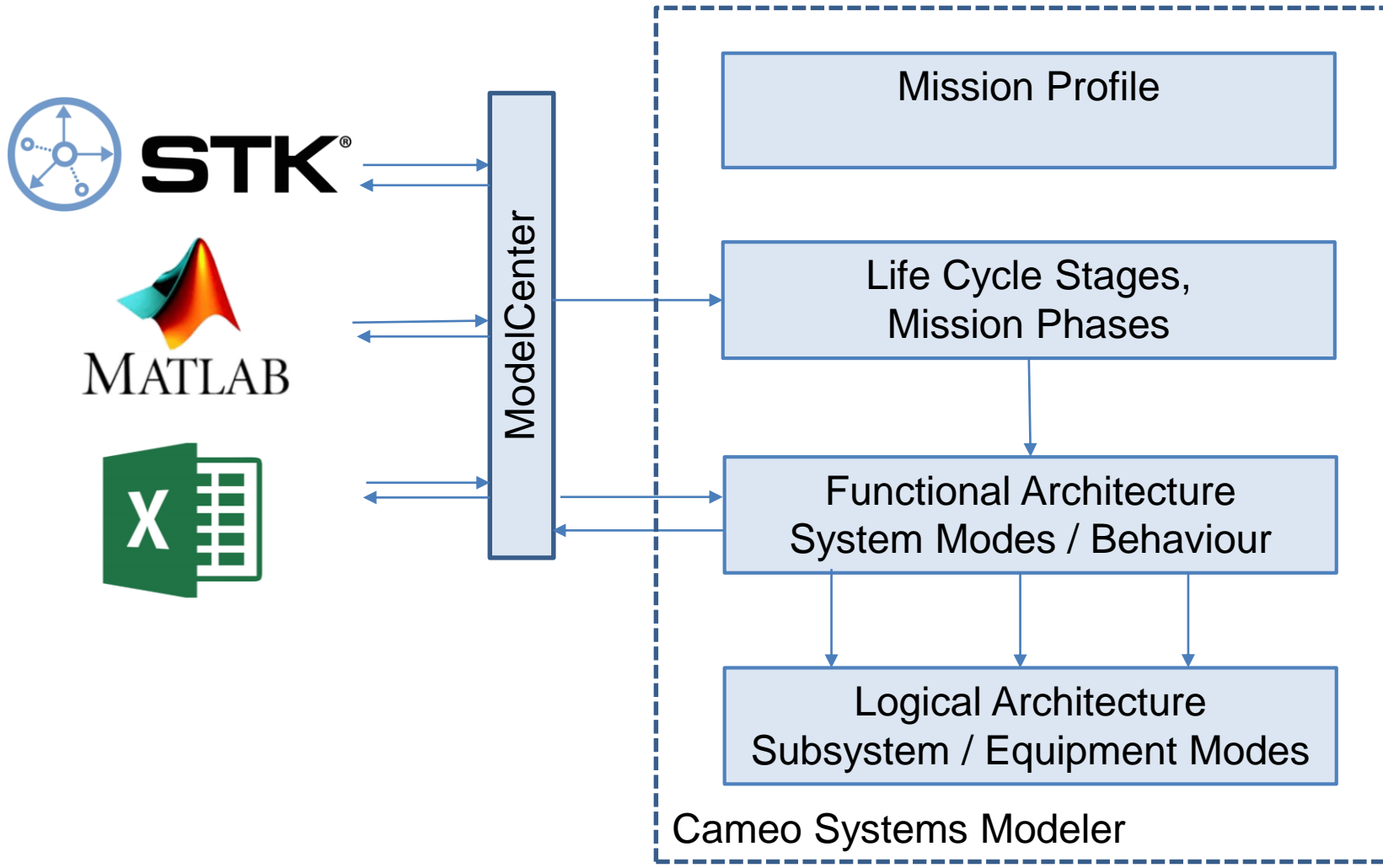
- **JUICE**
- Toulouse, France
- Science data allocation

eDeorbit is the largest effort so far – methodology and model template developed

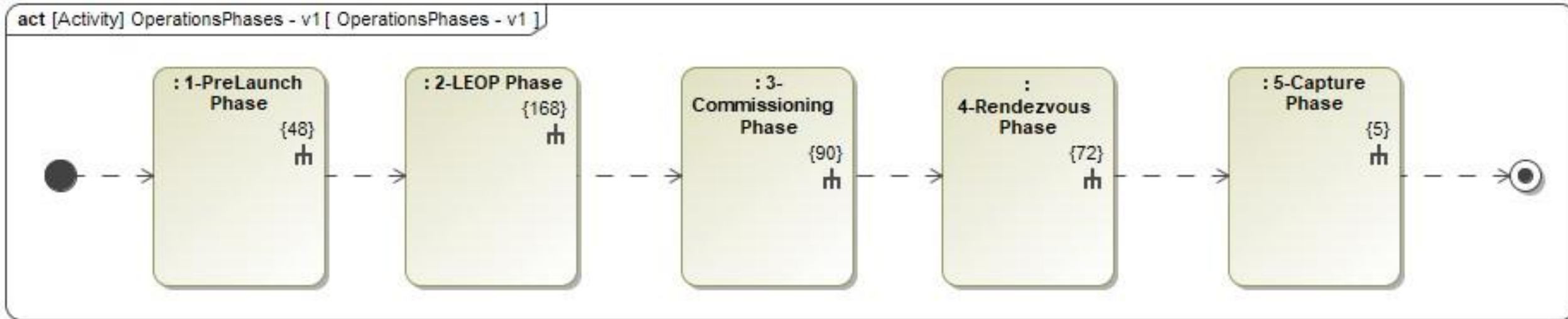
## eDeorbit

- Safety-critical rendezvous  
Deorbit Envisat
- Phase B1-B2  
Concept exploration  
System definition
- Developed using the  
'Federated and Executable  
Models' approach  
*Stephane Estable, Airbus*



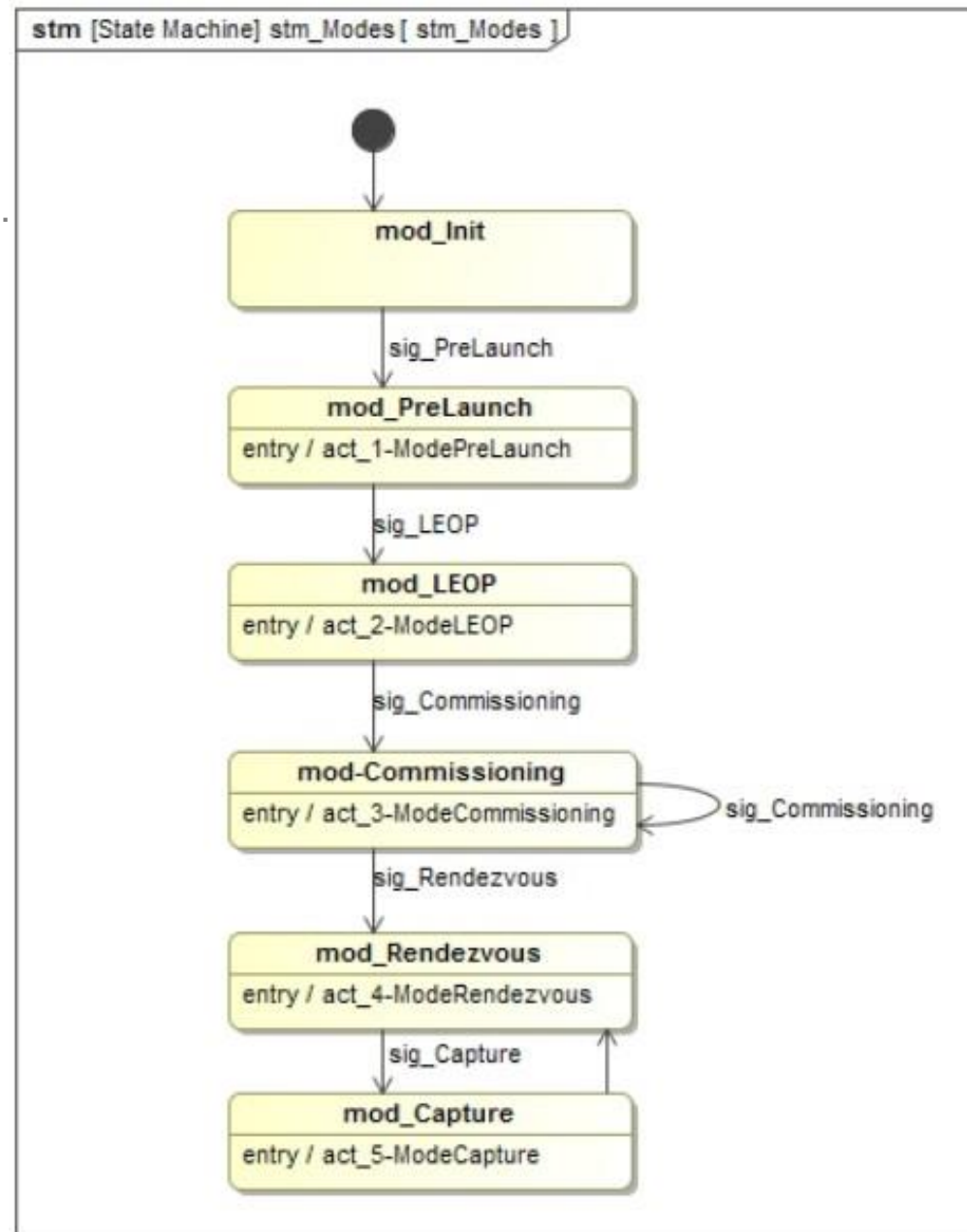


## eDeorbit - Phases

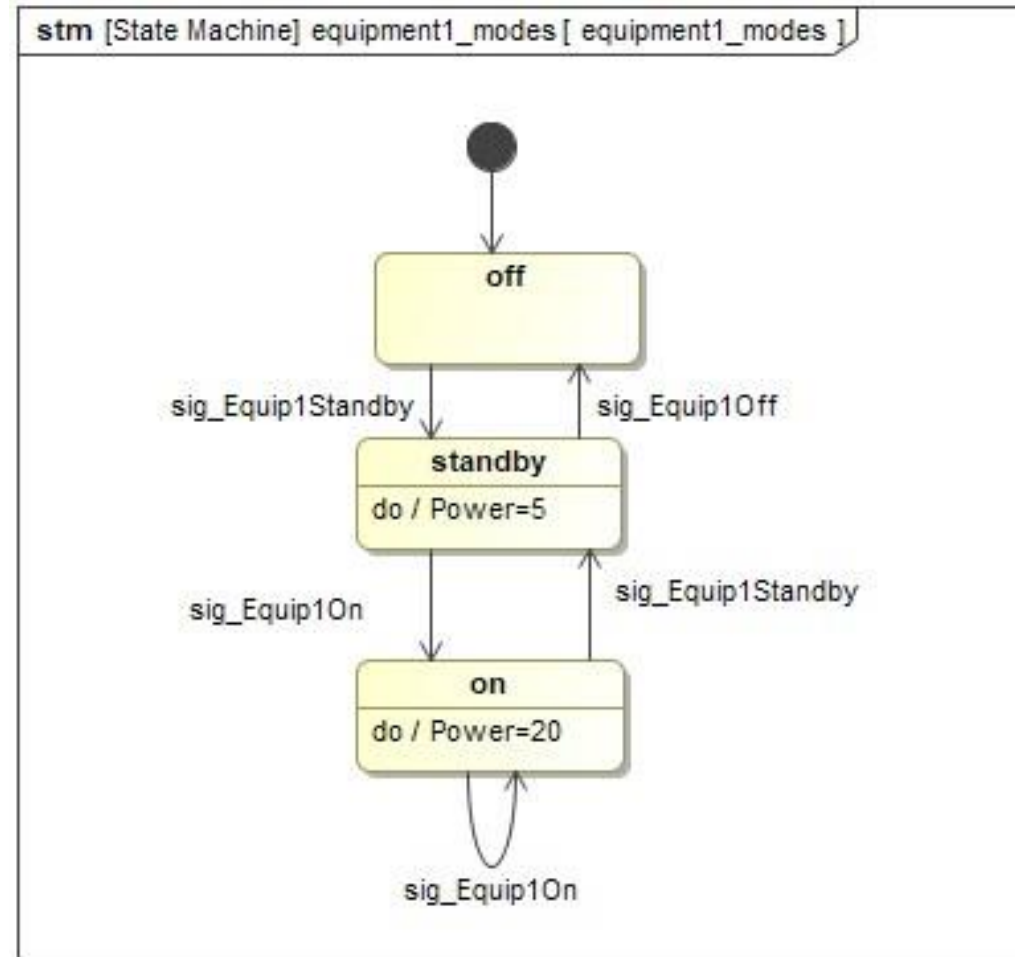




## eDeorbit - Modes



## eDeorbit – Subsystem (Modes)



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## Airbus Functional Avionics Interviews

Operations

Failure Detection, Isolation and Recovery (FDIR)

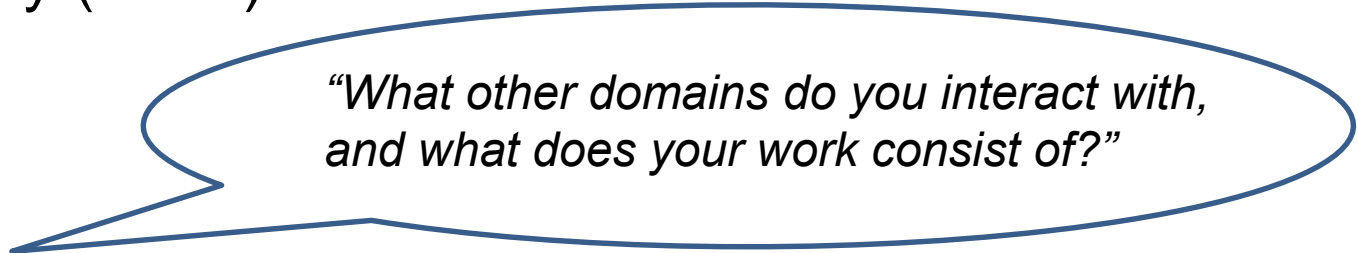
Software

Introductory Airbus interviews

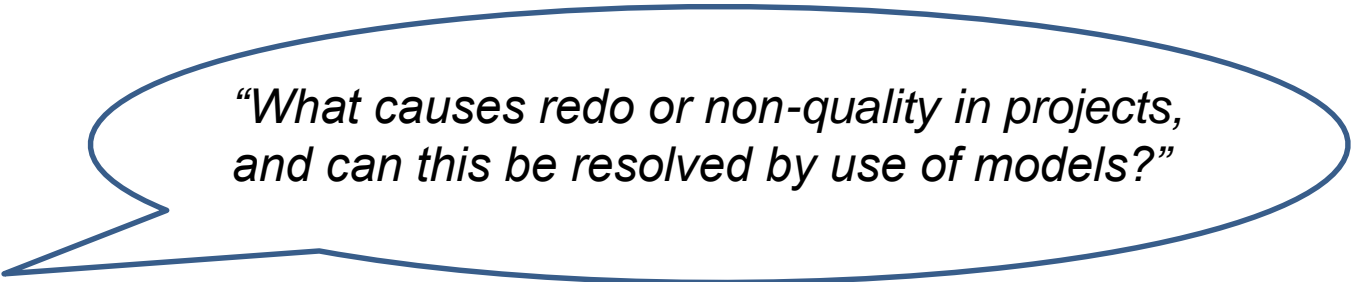
- Two sites (UK, France)
- 28 interviewees

Detailed Airbus interviews

- Three sites (UK, France, Germany)
- 23 interviewees



*“What other domains do you interact with, and what does your work consist of?”*



*“What causes redo or non-quality in projects, and can this be resolved by use of models?”*

## Airbus Functional Avionics Interviews - Outcomes

- 205 answers
- Process, Organisation, Tools
- Operations, FDIR, Software

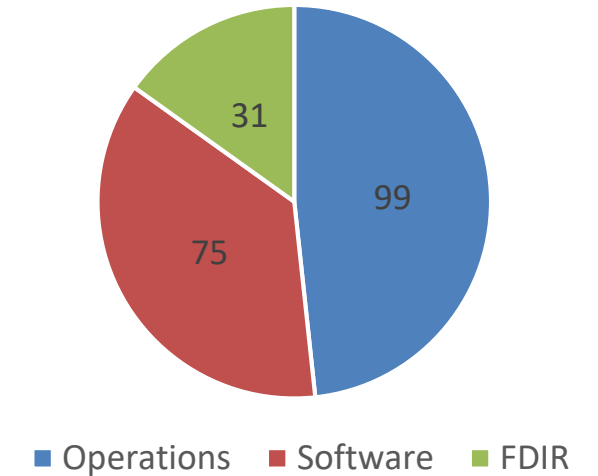
### Areas increasing the Cost of Non-Quality

- Interfaces between domains (e.g. SW – Ops)
- Lack of integrated toolset
- Lack of early validation of Concept of Operations

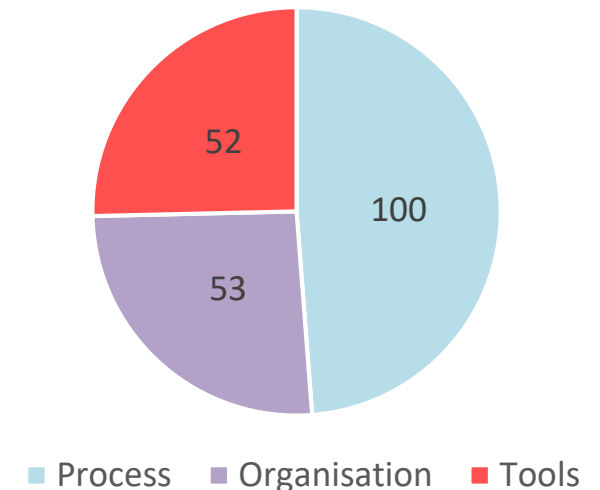
### Topics of interest

- **System-level early validation of Concept of Operations (ConOps)**
- **Model template development**

Responses from Domains



Response Categories



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## Airbus Interviews Publication

# The Long and Winding Road: MBSE Adoption for Functional Avionics of Spacecraft

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Airbus Defence and  
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Gunnels Wood Road  
Stevenage, UK  
SG12AS

- *Currently under review*

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## Modelling Objectives:

- Can we add simulation capabilities to the *Federated and Executable Models* approach?
- Can we do so without using ModelCenter (by Phoenix Integration)?
- Can we do so while maintaining the model's primary use a *System Description*

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## Mass Memory Sizing

### *Biomass Spacecraft*

#### Instrument (Radar)

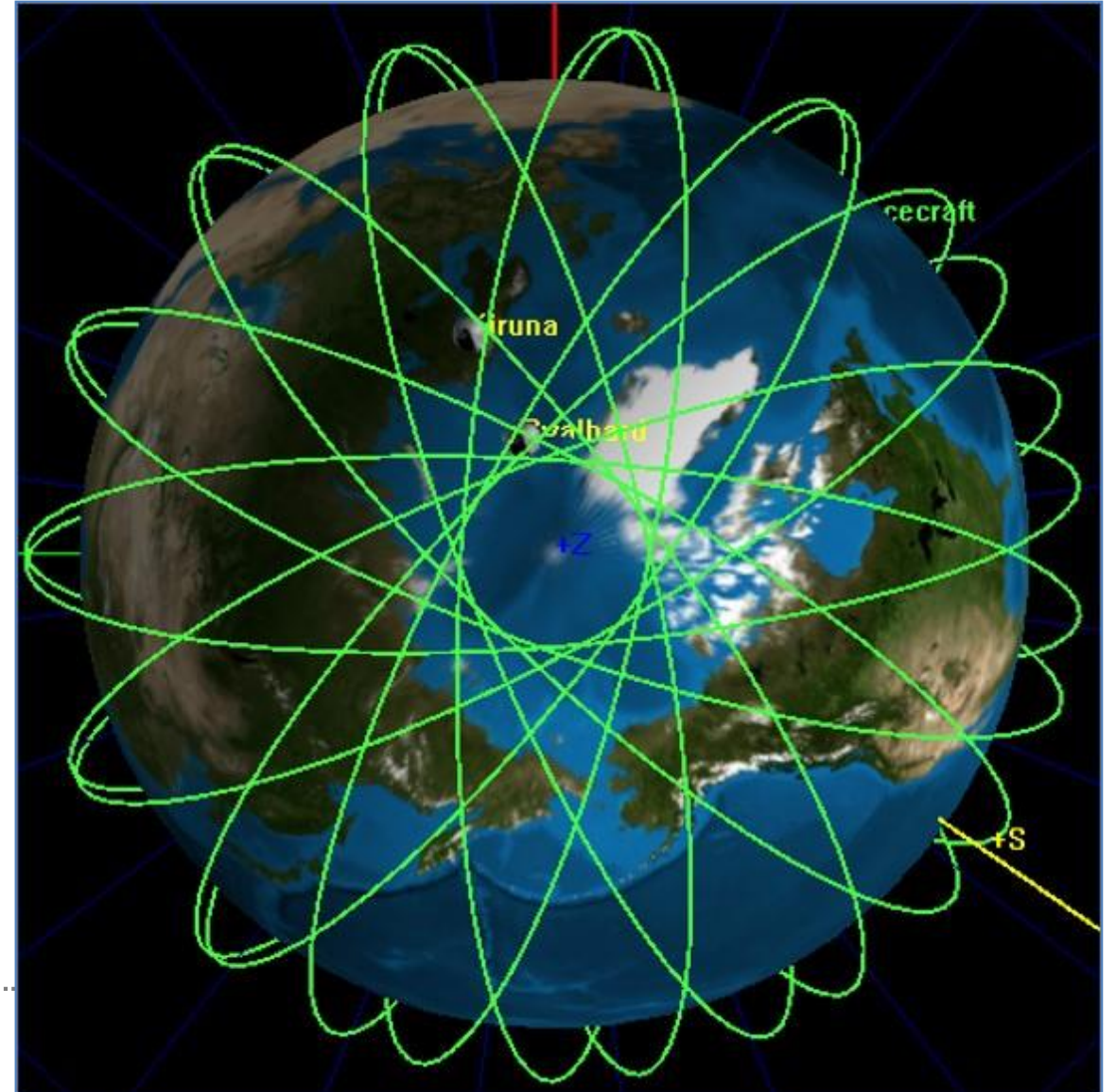
- On over land
- Off over ocean

#### Antenna

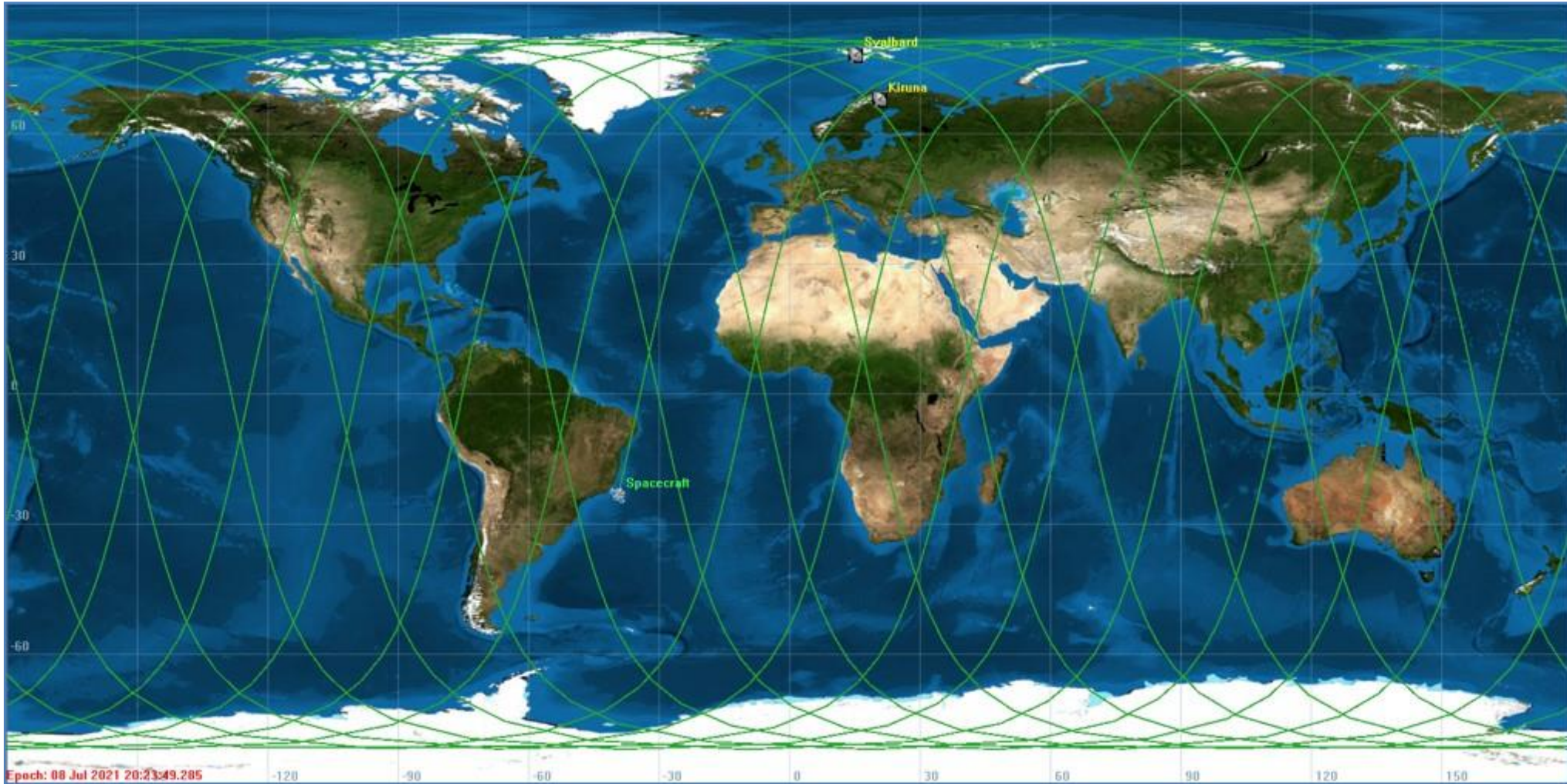
- On over ground station
- Off when not over ground station

#### Memory Unit – 3x Directories

- Reading (when Antenna on: downlinking)
- Writing (when Instrument on: recording)
- Both (when Antenna and Instrument on)





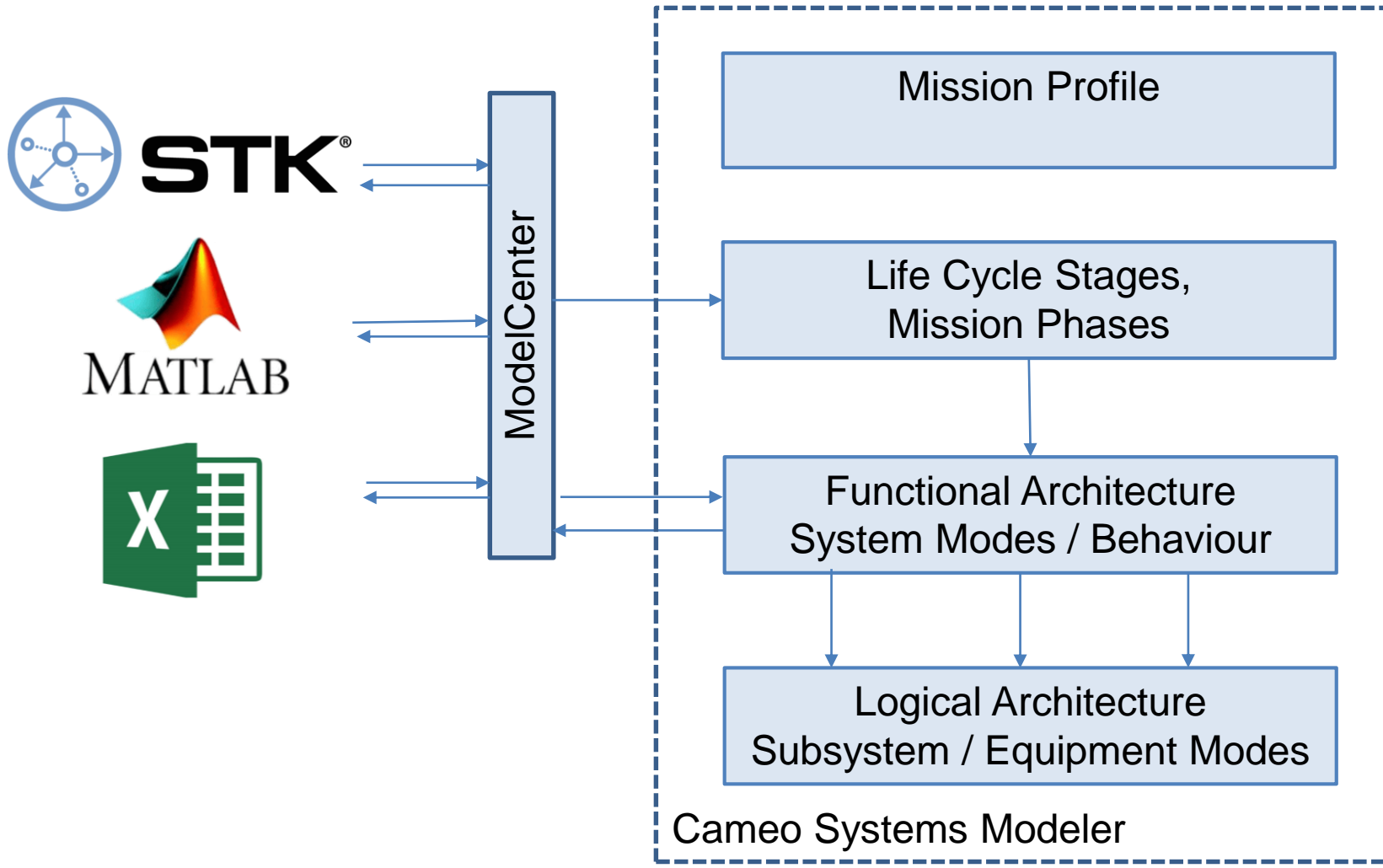


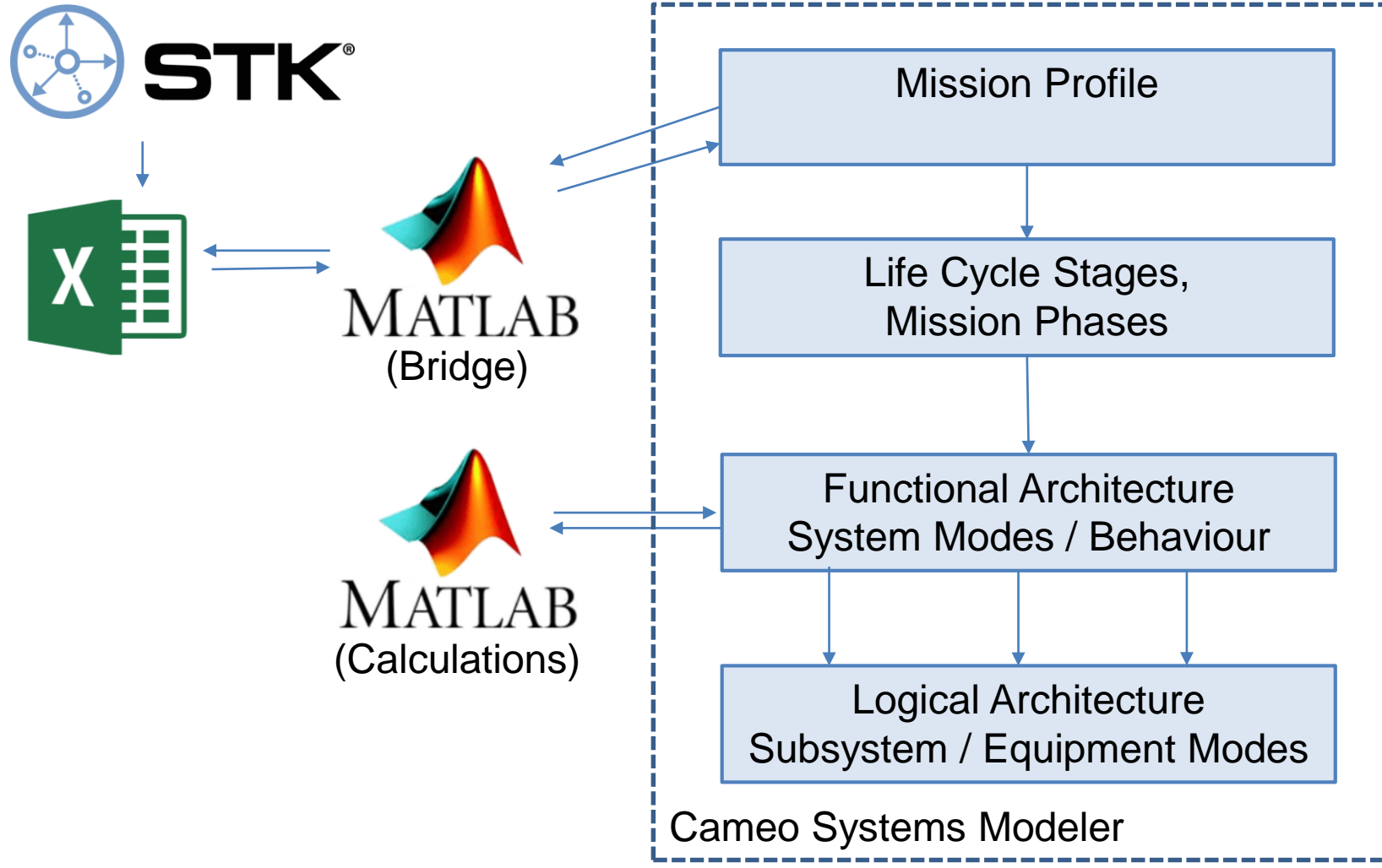


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## Mass Memory Use Case Objectives

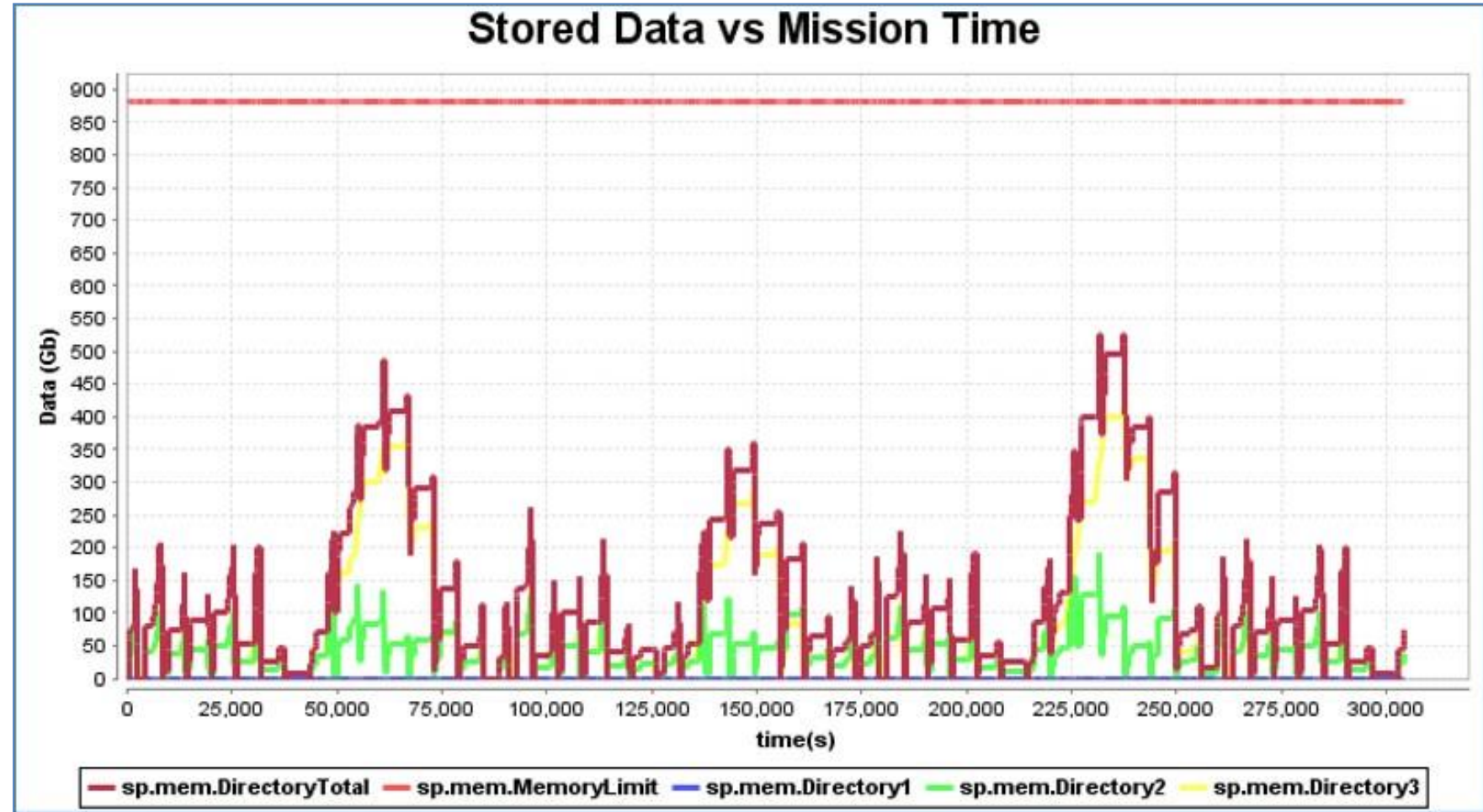
1. Validate spacecraft memory unit design
  - Validate total memory allocation (880Gb)
  - Optimise Memory Unit directory sizes
  - Validate model against Excel results





## Mass Memory Use Case Results

1. Memory allocation
  - Full simulation results show a ~40% margin
2. Optimised directory sizes
3. Validated model against subcontractor Excel data

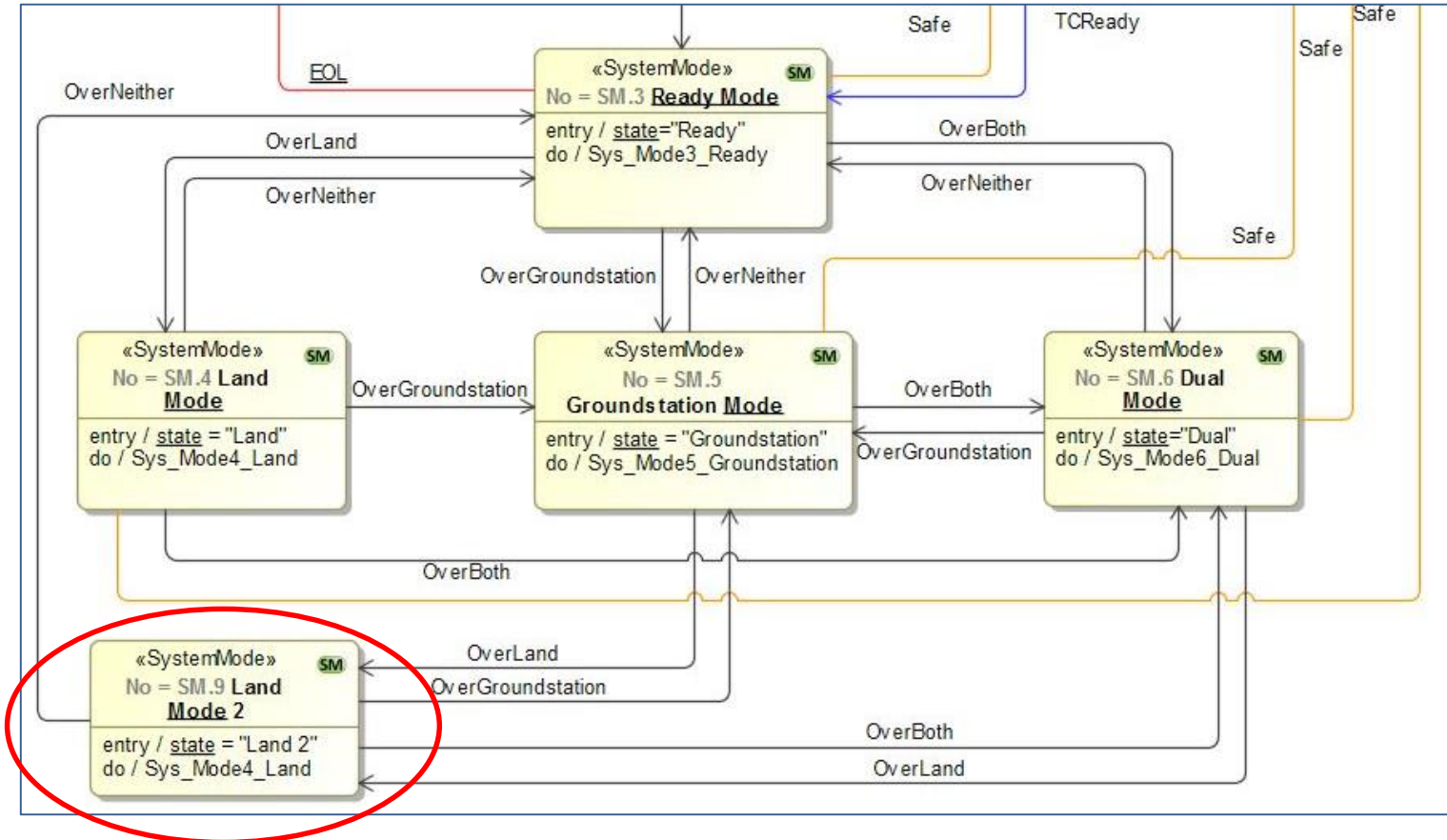


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## Biomass Model Flexibility

1. Changes to Requirements
2. Additional Mode
3. Change no. of Directories
4. Contingency Analysis
5. Mission Updated

## E.g. Additional Mode



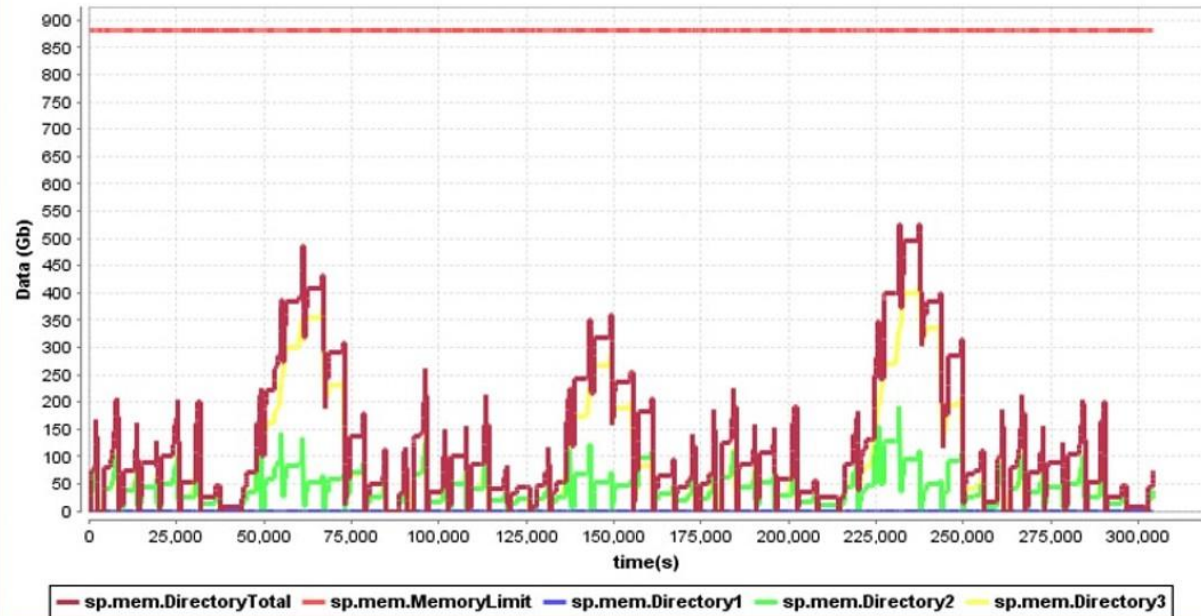
Introduce another mode

- Increased Land Mode
- Increased capability (4 x rate)
- Effective after downlink
- Observe effect on mass memory
- Effectively changing system response

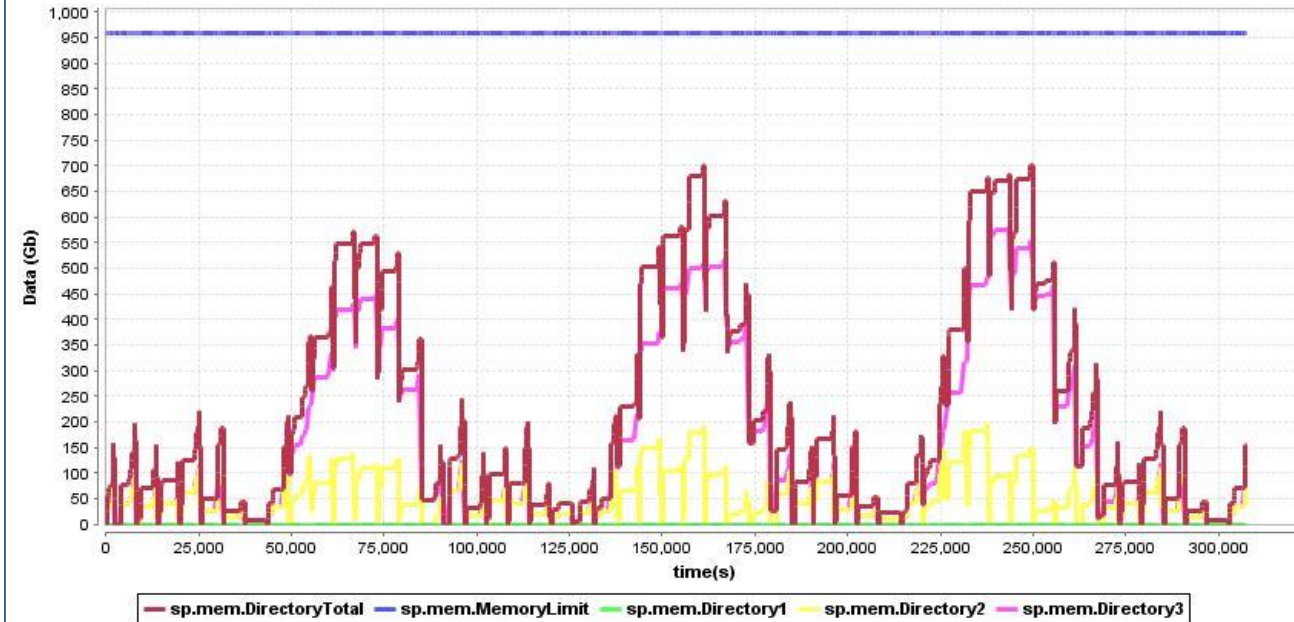


## E.g. Additional Mode

Stored Data vs Mission Time



Stored Data vs Mission Time



Mission not changed, System response has been updated  
Effects immediately observable

**Time Taken:**  
Add Mode in Mode Diagram: *minutes*  
Produce Mode Activities: *~day*

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## Mass Memory Use Case Outcomes – in terms of Template

- Demonstrated key features of the model structure
  - Maintained separation of *Mission* and *System*  
Allocated system functionality to system logical architecture
- Added simulation capabilities to the original model – *Federated and Executable Models*
  - Mass memory sizing calculations
  - Removed ModelCenter – uses MATLAB as bridge to connect to Excel, AGI STK
- *Mission Operations Concept Document* to *Mission Operations Concept Model*
  - Maintained the model's primary purpose as a system description
  - The underlying structure that allows the model to be executed can be seen as a *bonus*



## Mass Memory Sizing Use Case Publications





Using MBSE Techniques to perform Early Validation on the Data Handling Unit of an Earth-Observation Spacecraft

22<sup>nd</sup> May 2018

Joe Gregory, Lucy Berthoud, Antonio Prezzavento, Joanna O'Rourke, Stephane Estable, Alexandre Cortier

University of Bristol, funded by Airbus

Platinum Sponsor:  Gold Sponsors:       

*Early Validation of the Data Handling Unit of a Spacecraft Using MBSE*

IEEE Aerospace Conference. Yellowstone Conference Center, Big Sky, Montana  
4<sup>th</sup> Mar 2019

Joe Gregory, Lucy Berthoud, Theo Tryfonas      University of Bristol, UK  
Antonio Prezzavento      Airbus, UK

bristol.ac.uk

**Early Validation of the Data Handling Unit of a Spacecraft Using MBSE**

<p>Joe Gregory Department of Aerospace Engineering Queens Building University Walk University of Bristol, UK BS81TR joe.gregory@bristol.ac.uk</p>	<p>Lucy Berthoud Department of Aerospace Engineering Queens Building University Walk University of Bristol, UK BS81TR lucy.berthoud@bristol.ac.uk</p>	<p>Theo Tryfonas Department of Civil Engineering Queens Building University Walk University of Bristol, UK BS81TR theo.tryfonas@bristol.ac.uk</p>	<p>Antonio Prezzavento Airbus Defence and Space Gunnels Wood Road Stevenage, UK SG12AS antonio.prezzavento@airbus.com</p>
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**Investigating the Flexibility of the MBSE Approach to the Biomass Mission**

<p>Joe Gregory Department of Aerospace Engineering Queens Building University Walk University of Bristol, UK BS81TR</p>	<p>Lucy Berthoud Department of Aerospace Engineering Queens Building University Walk University of Bristol, UK BS81TR</p>	<p>Theo Tryfonas Department of Civil Engineering Queens Building University Walk University of Bristol, UK BS81TR</p>	<p>Antonio Prezzavento Airbus Defence and Space Gunnels Wood Road Stevenage, UK SG12AS</p>	<p>Ludovic Faure Airbus Defence and Space Gunnels Wood Road Stevenage, UK SG12AS</p>
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## Critical Sequences

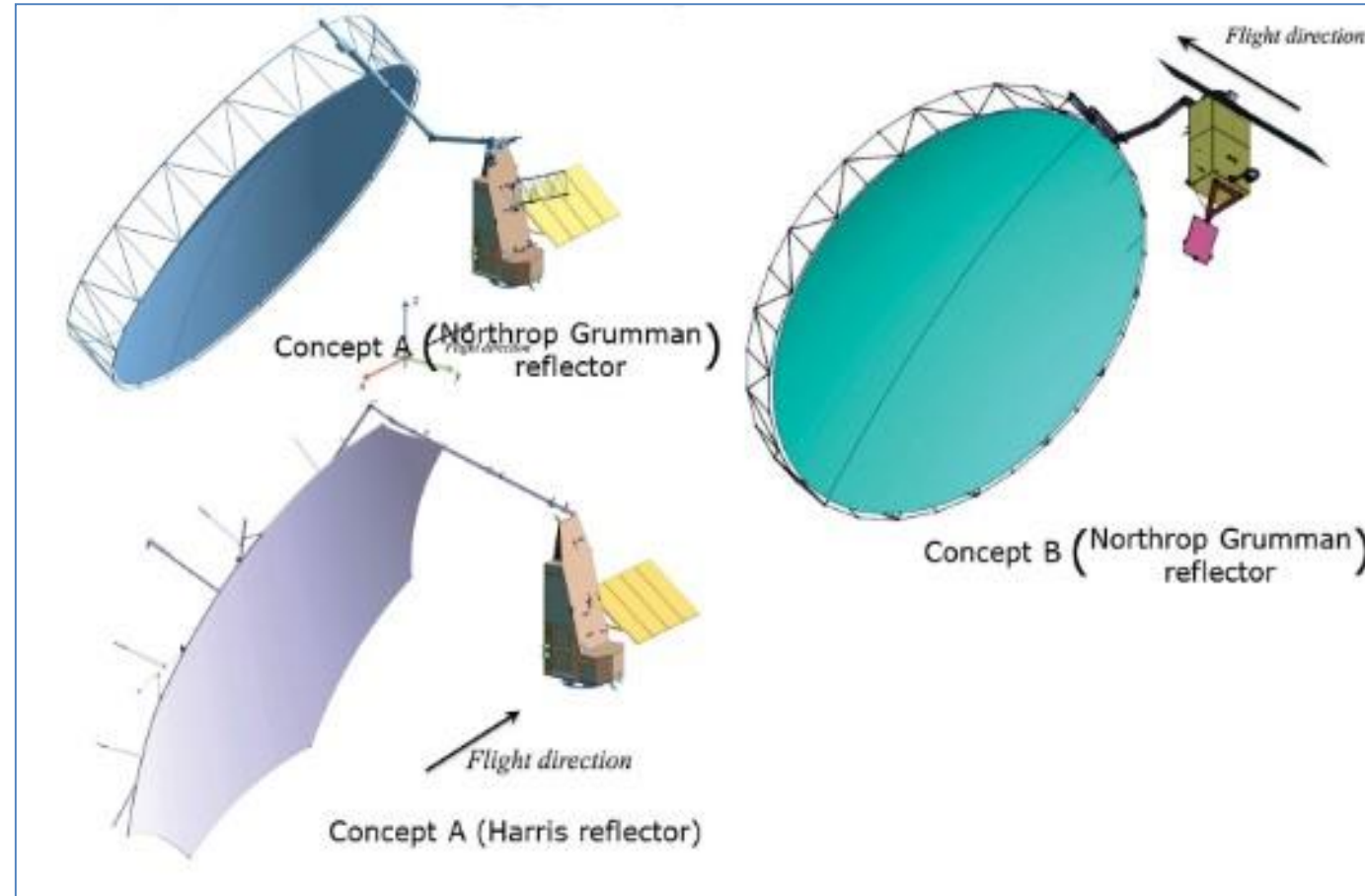
### *Biomass Spacecraft*

#### Deployable Reflector

- Deployed soon after launch
- Critical procedure
- Must be in contact with ground

#### Considerations

- Communication with ground
- Modes / states of system, subsystems and equipment
- Multiple decision nodes

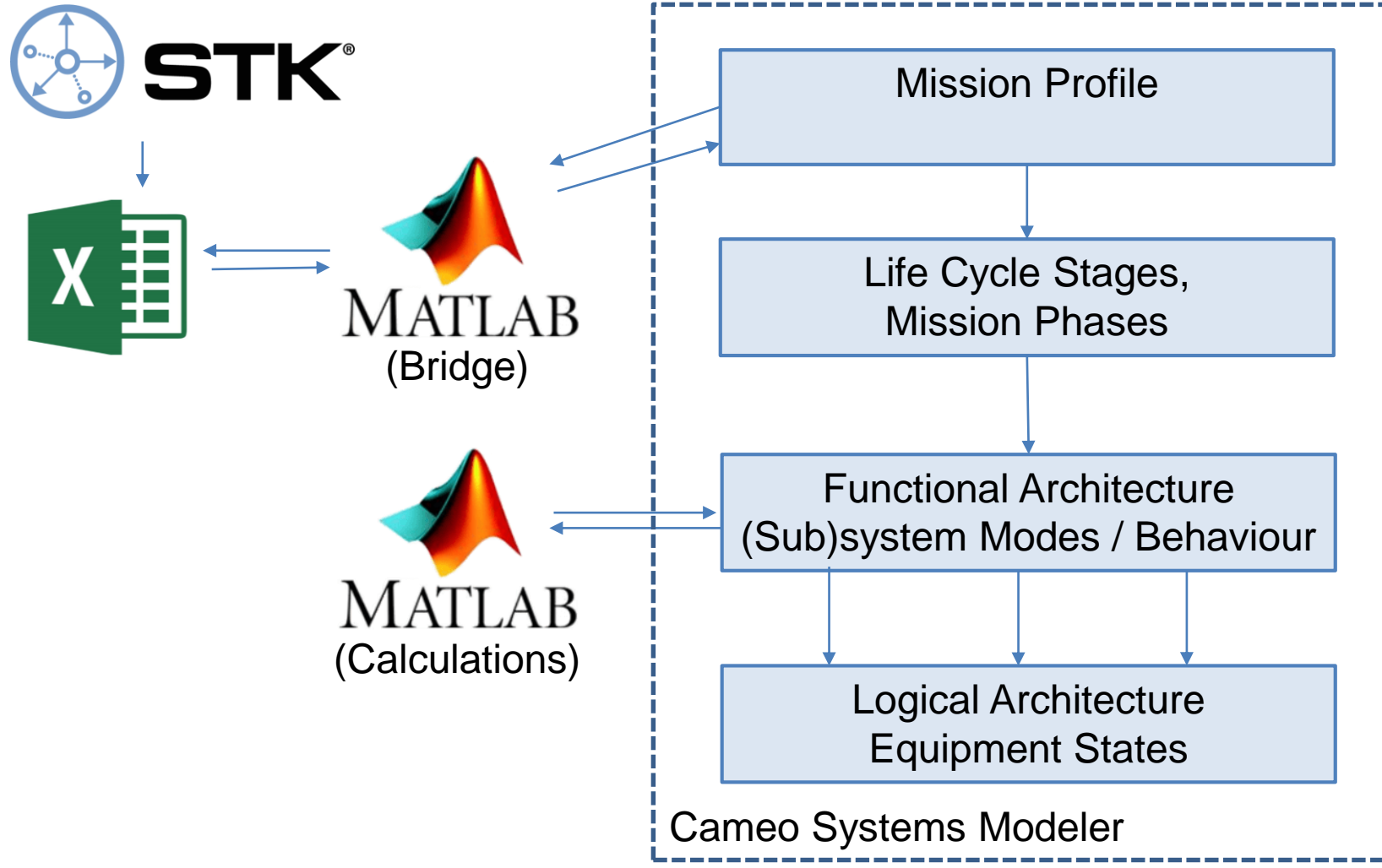


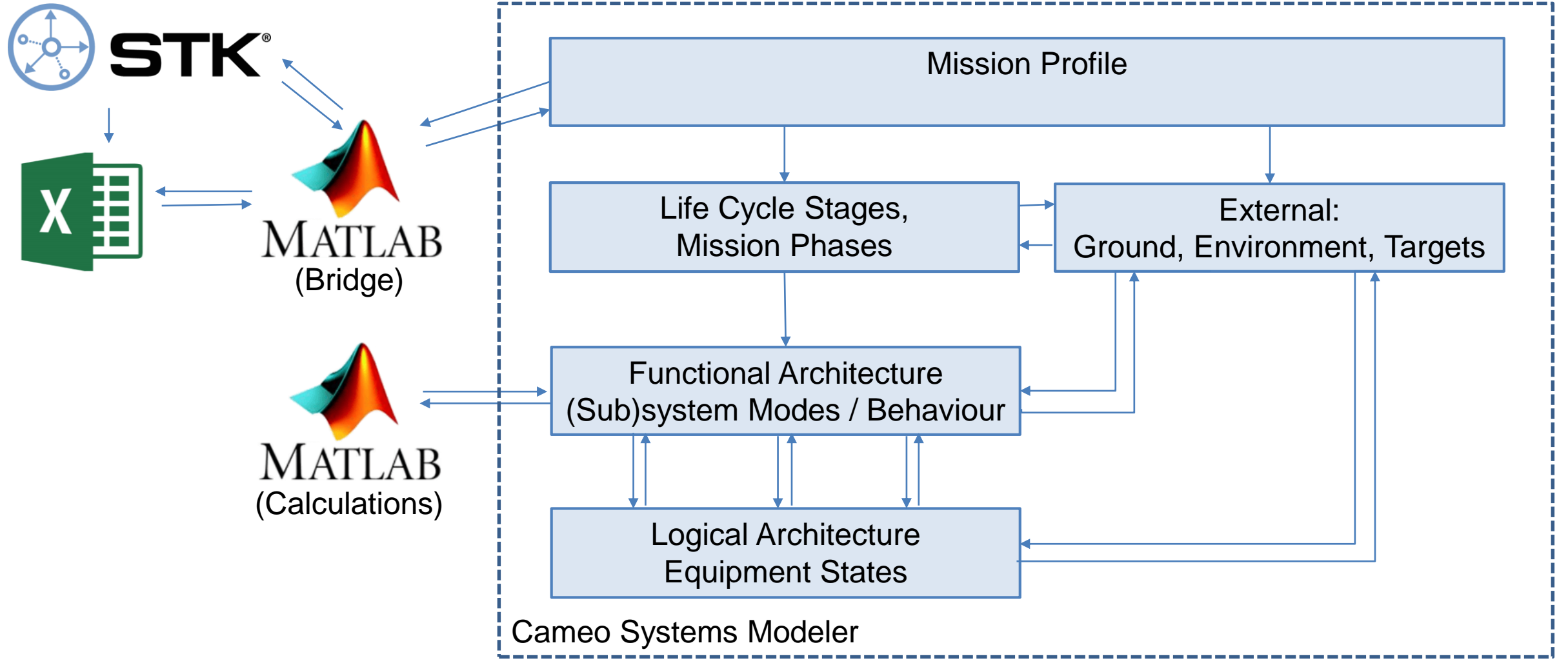
F. Hélière, F. Fois, M. Arcioni, P. Bensi, M. Fehring, and K. Scipal, "Biomass P-band SAR interferometric mission selected as 7th Earth Explorer Mission," in *Proceedings of 10th European Conference on Synthetic Aperture Radar, EUSAR 2014*

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## Critical Sequences Use Case Objectives

1. Validate the reflector deployment sequence
  - Ensure that deployment sequences fully describe all possible scenarios.
  - Demonstrate flexibility of the system by postponing launch, altering orbit, etc.
  - Demonstrate that all system modes, subsystems modes and equipment states have been described.
  - Begin assigning functions to logical architecture







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## Critical Sequences Use Case Results

Not quite there yet...



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## Critical Sequences Use Case Outcomes – in terms of Template

*Outcomes so far...*

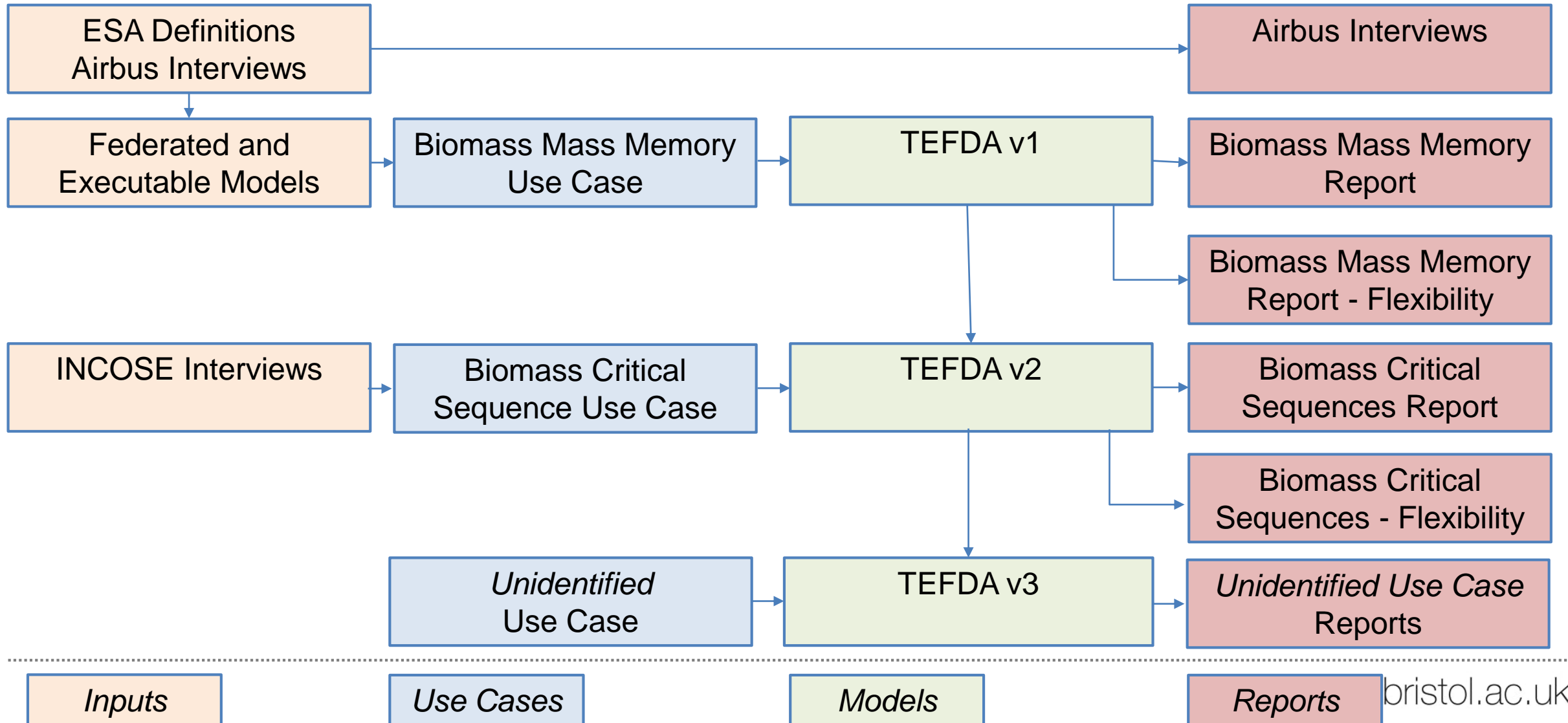
- Can run STK from Cameo Systems Modeler (with *Katy Pugh, University of Bristol*)
  - Previously limited to reading Excel-based results into Cameo Systems Modeler.
  - Now can do the following:
    1. Define the parameters of an STK scenario in Cameo Systems Modeler
    2. Open STK and run the scenario from Cameo Systems Modeler
    3. Generate results in STK and output these to Cameo Systems Modeler
- Made behaviour *Modular*
  - E.g. added 'External' block - Can now add external behaviour (e.g. Environment, Targets)
  - Can follow patterns to add system behaviour and choose which to include in simulation

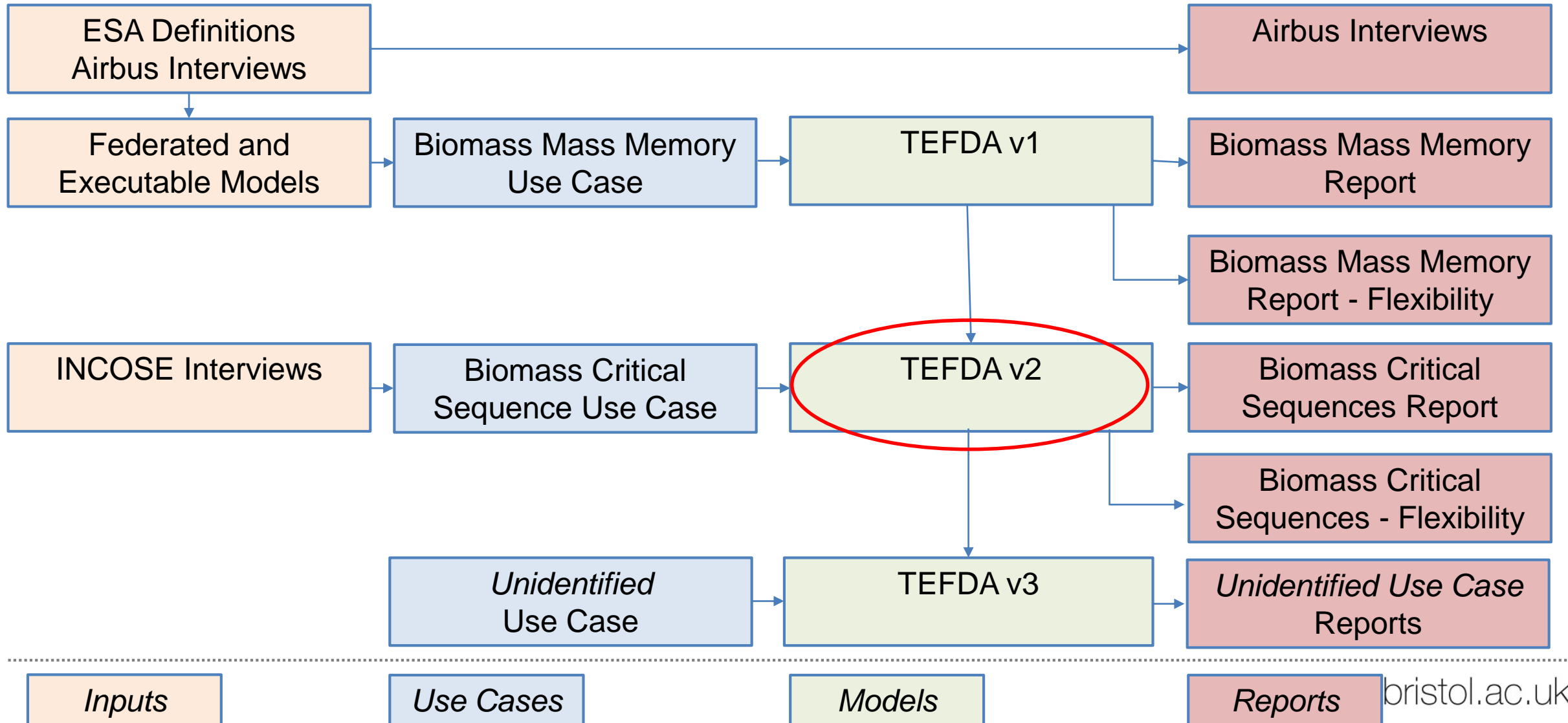


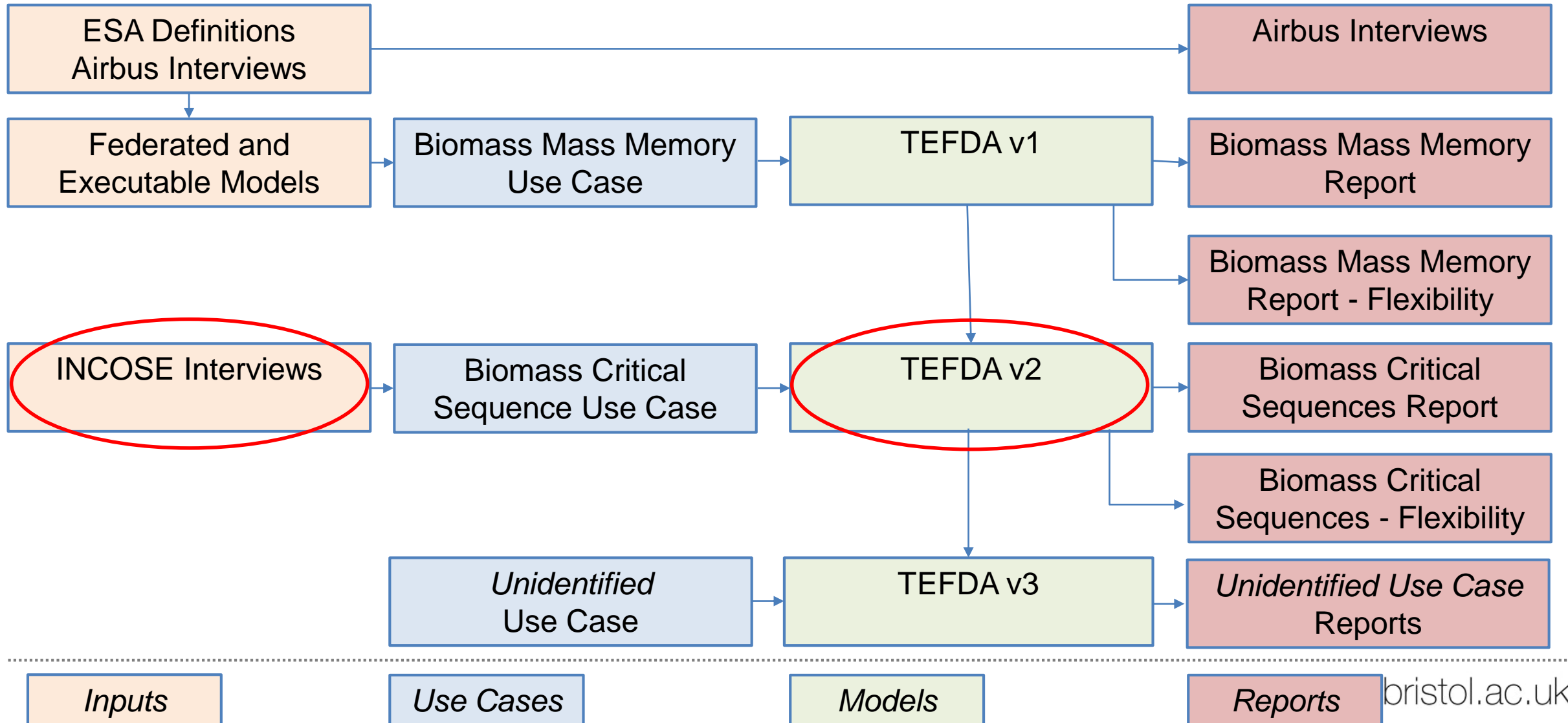
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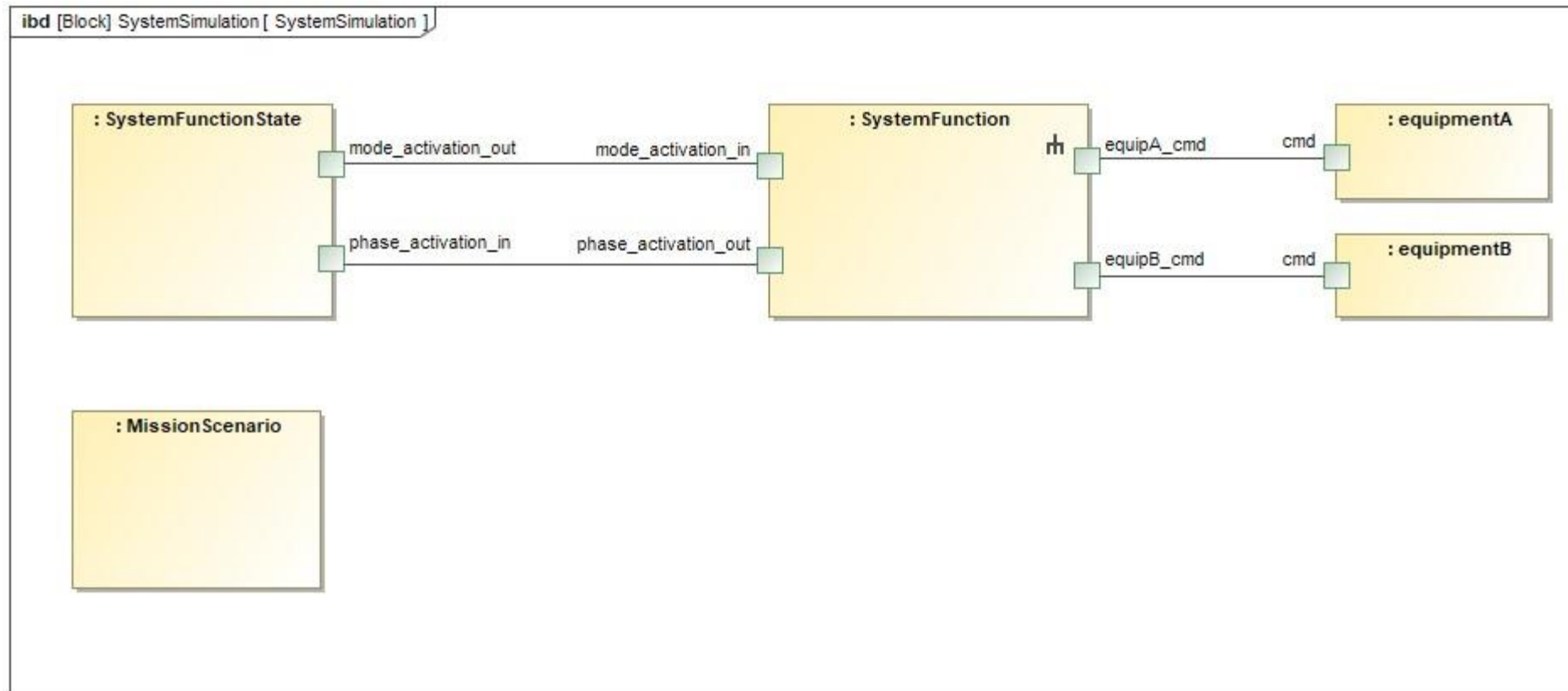
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## Can we generalise and produce a model template for use on other missions?

In Progress!

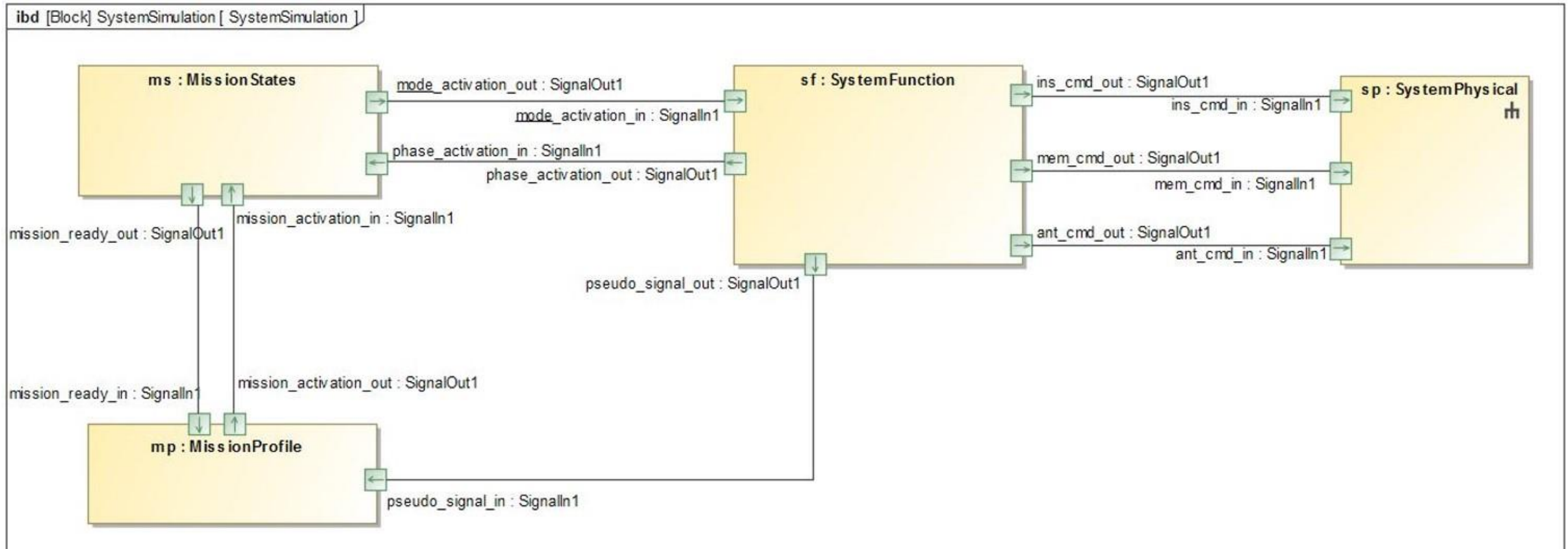
- **Manage complexity**
- Separate *descriptive* and *analytical* aspects of the model
- Measure success

## Top-Level Internal Block Diagram - Initial

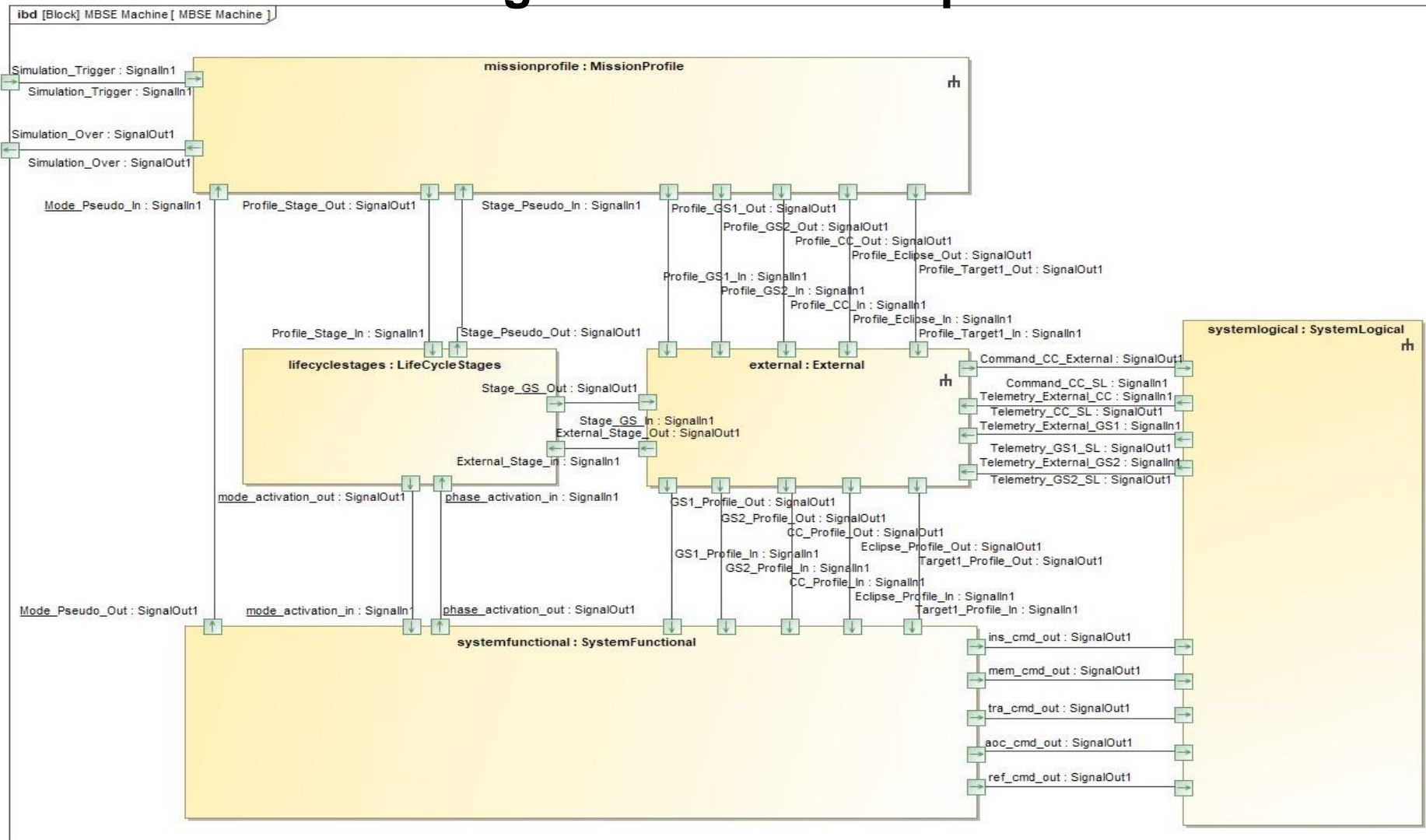




## Top-Level Internal Block Diagram – Mass Memory Use Case



## Top-Level Internal Block Diagram – Critical Sequences Use Case

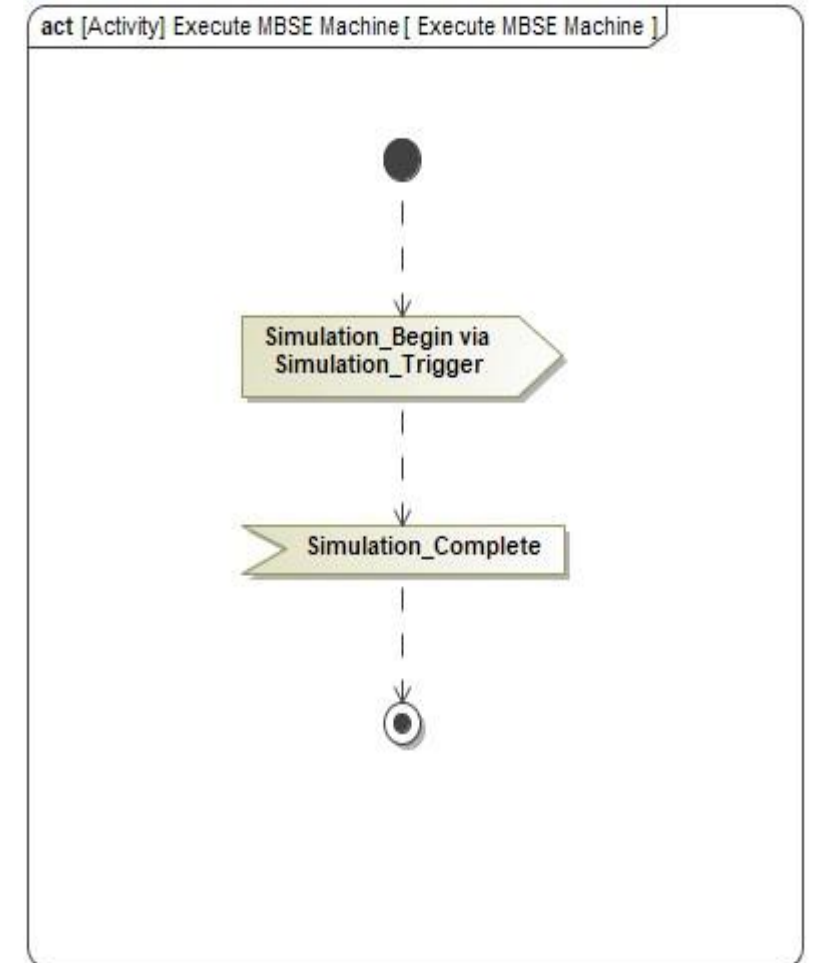
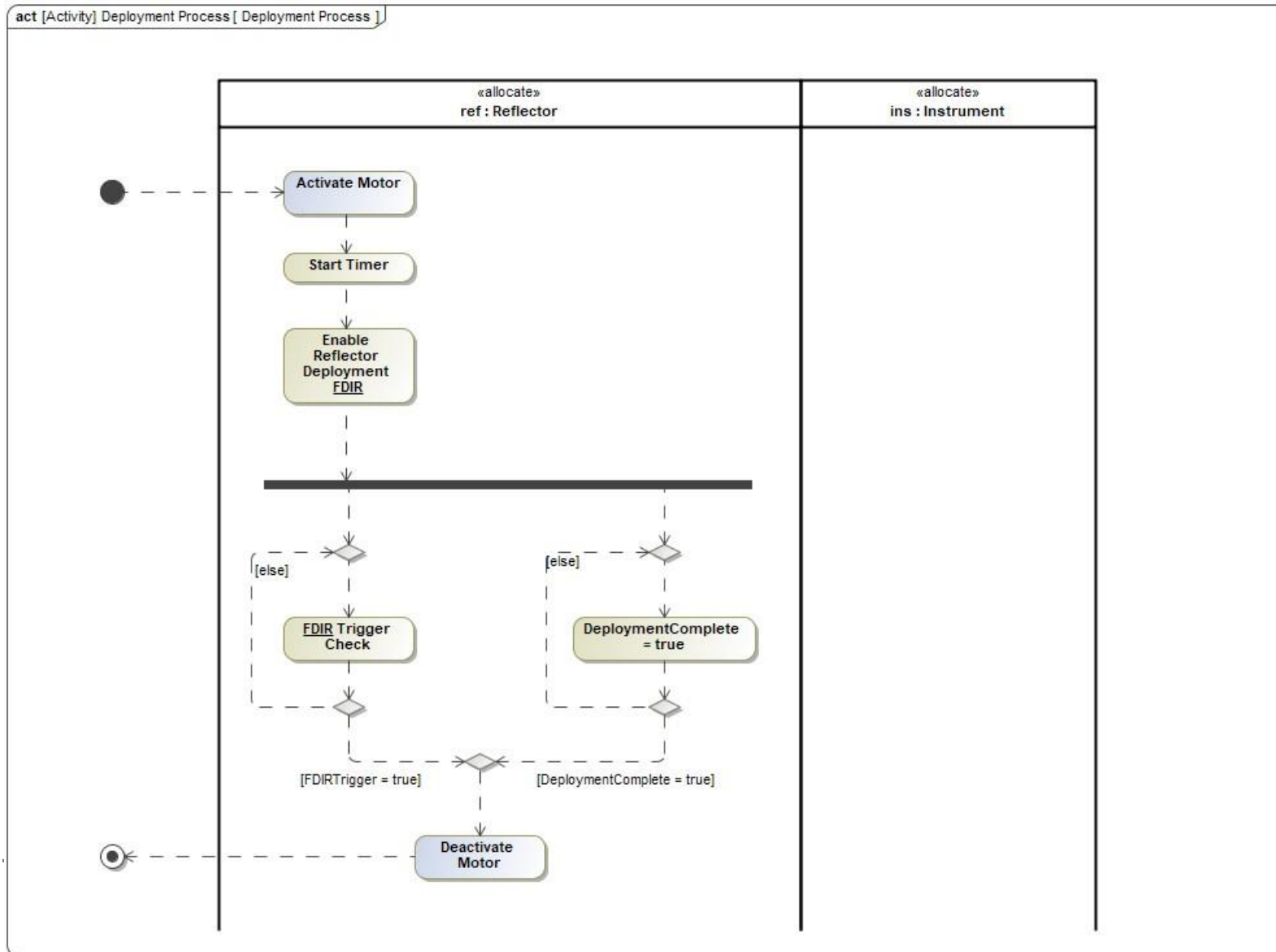


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## Can we generalise and produce a model template for use on other missions?

In Progress!

- Manage complexity
- **Separate *descriptive* and *analytical* aspects of the model**
- Measure success



---

## Can we generalise and produce a model template for use on other missions?

In Progress!

- Manage complexity
- Separate *descriptive* and *analytical* aspects of the model
- **Measure success**

---

## Measuring Success

### Possible Metrics

- Communicability
- Consistency
- Traceability
- Reusability
- Accuracy

Possibility of MBSE / DBSE experiment to come



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Thank you for listening

Questions / Feedback  
gratefully received!